

SoCWire: a SpaceWire inspired fault tolerant Network on Chip approach for reconfigurable System-on-Chip in Space applications

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Outline

- Data Processing Units based on Xilinx Virtex FPGA
- Future space missions
- Advantage: in-flight and dynamic reconfigurability
- Reconfigurability in space: requirements
- Bus structure based dynamic reconfigurable SoC
- Network-on-Chip approach: SoCWire
- Dynamic reconfigurable architecture



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Data Processing Units (DPUs) in space

Data Processing Unit:

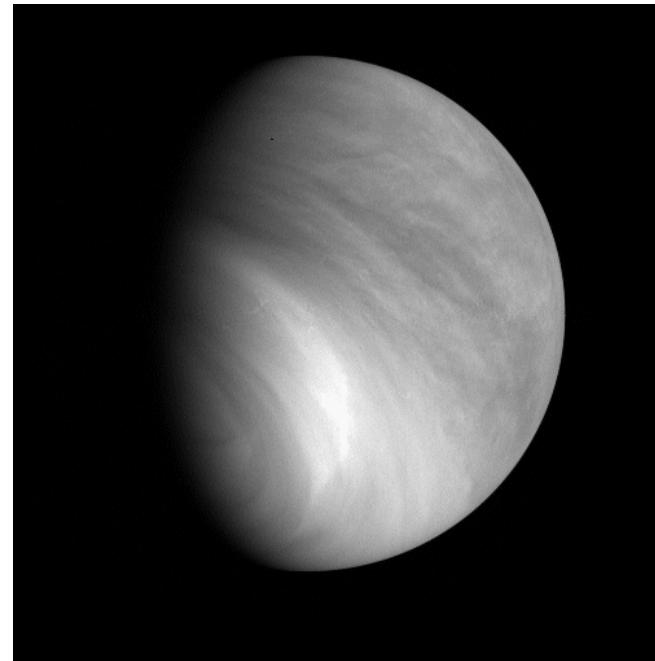
- are used as interface between spacecraft and several instrument sensor electronics or heads, providing the operational control and specific data processing of scientific space instruments.

Venus Monitoring Camera (VMC)

- Based on Xilinx Virtex SRAM-cell FPGA
- LEON-2 VHDL model 20 MIPS
- Data rate 14Mbps
- Image processing (compression) in software
- High availability
- Very flexible
- reconfigurability have been used only during the development phase on ground.

Xilinx Virtex FPGAs

- Support dynamic partial reconfiguration

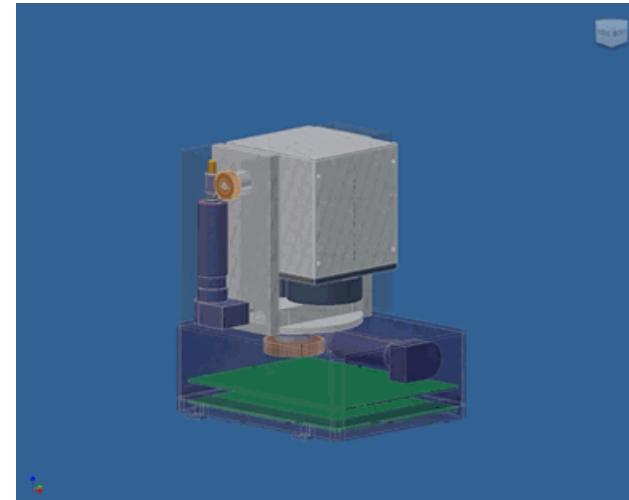
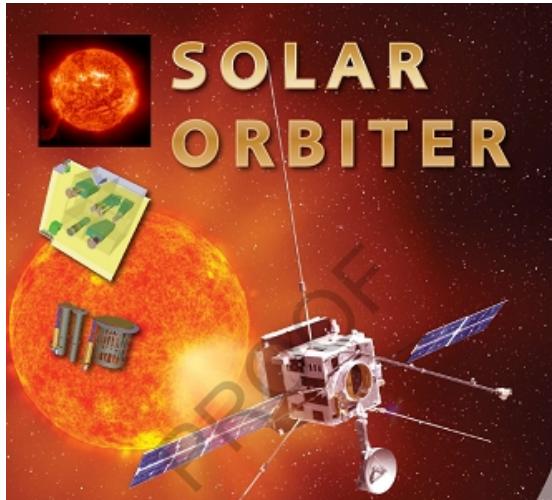


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Future Space Missions



PHI on Solar Orbiter (ESA)

- 4Mpixel detector
- data rate up to 500 Mbps

LEVIS

proposed by IDA for Lunar Exploration Orbiter (LEO)

- 5Mpixel detector
- Image acquisition and 1080p HDTV Video (30 Frames/s)
- data rate up to 750 Mbps

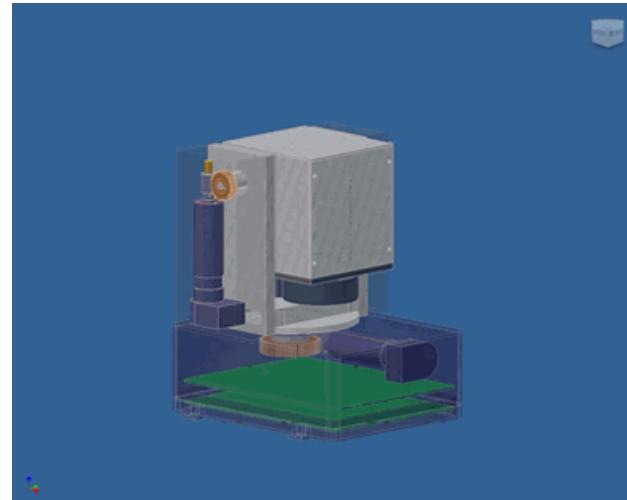
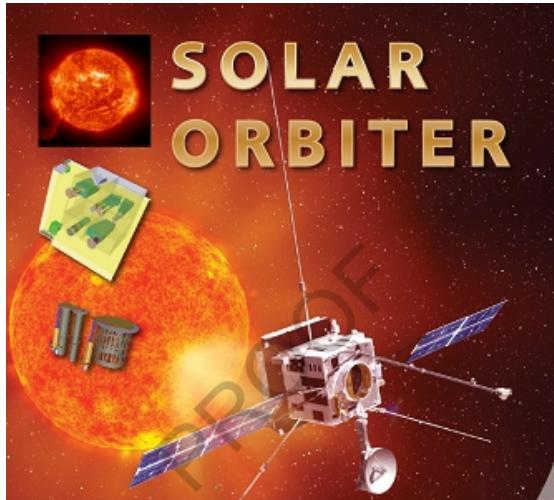


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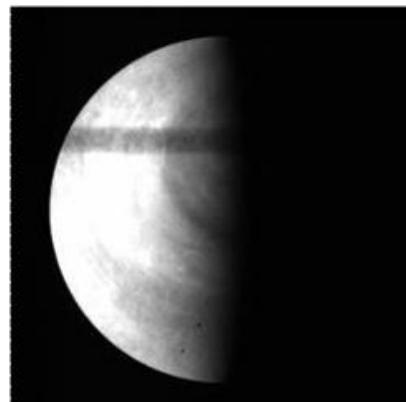
Future Space Missions



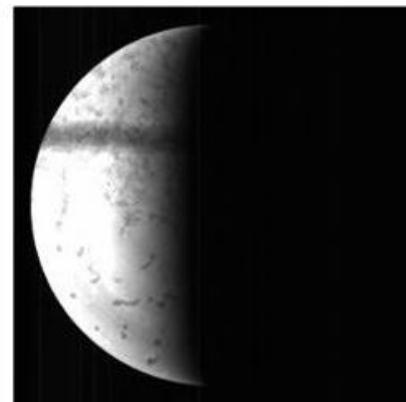
- Data rate to S/C 20..60 kbps
- Future Space mission demand high-performance on-board processing
- Classical ground processing steps like measurement data inversion and subsequent data evaluation need to be performed on-board

Adaptable on-board processing

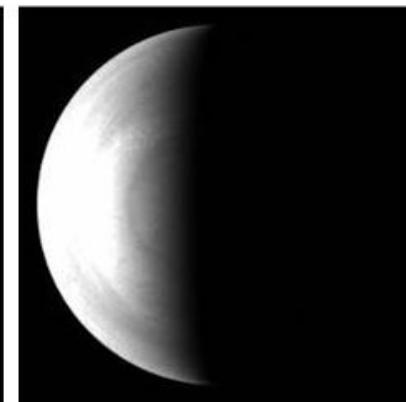
UVB Channel 2006



UVB Channel 2009



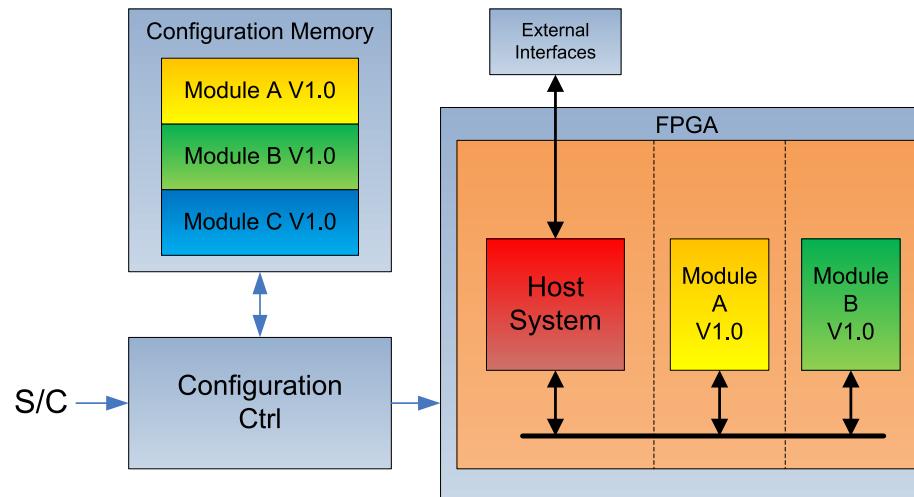
Flat-field UVB 2009



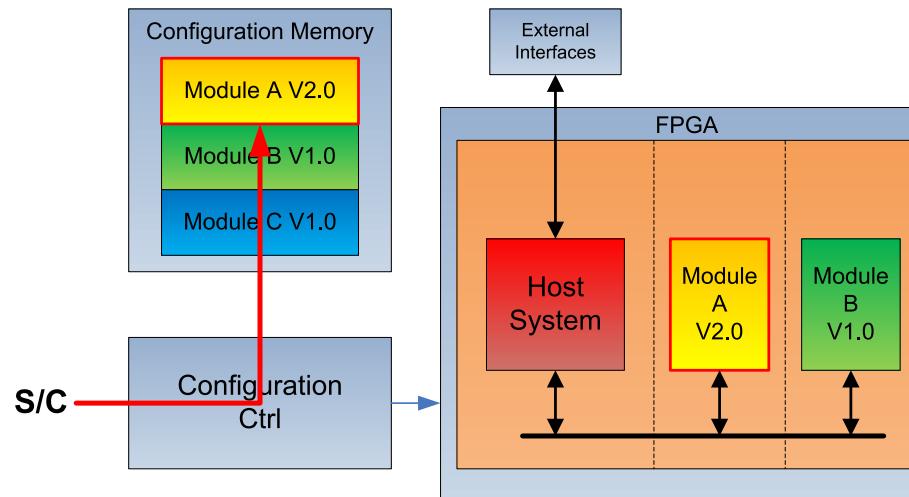
VMC CCD anomaly

- Requirement by the scientists: On-board processing must be adaptable to mission specific requirements

Advantage: In-flight and Dynamic reconfigurability

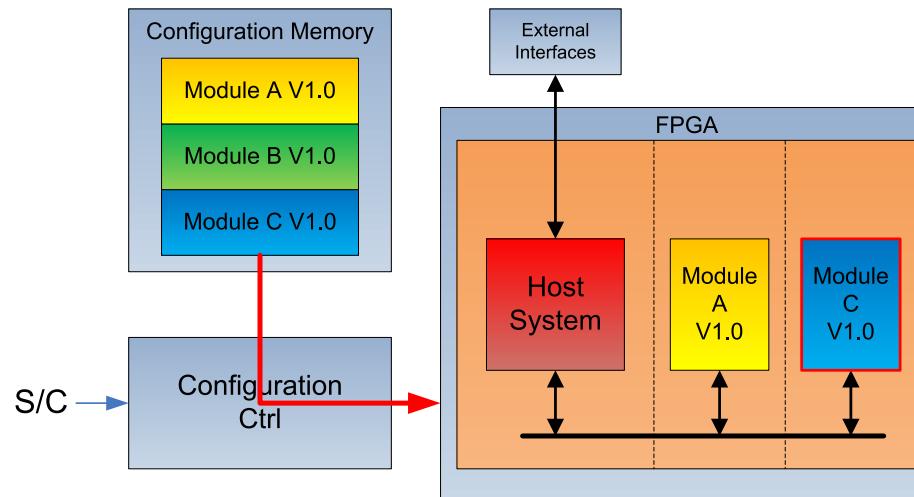


Advantage: In-flight and Dynamic reconfigurability



- Update of processing modules
- + **Improvement: maintenance and functionality**
- + **Especially valuable for space applications with non-accessibility of maintenance points**

Advantage: In-flight and Dynamic reconfigurability



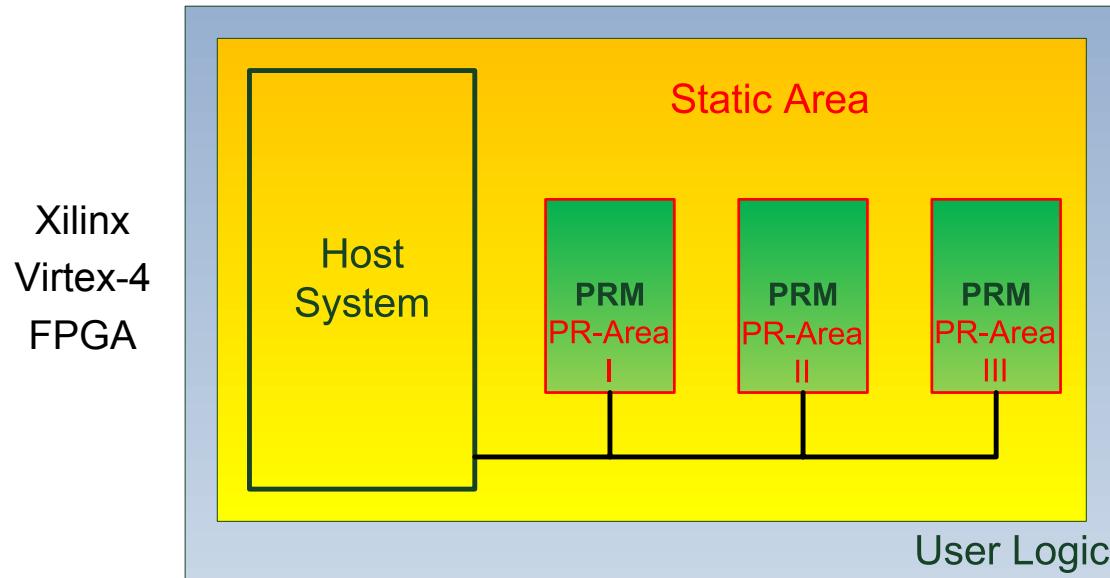
- Dynamic partial reconfiguration of modules
 - + **run-time adaptive functionality**
 - + **reduces resource utilization and power consumption**

Requirements for in-flight reconfigurability and dynamic reconfiguration

- On ground achieved qualification of the system has to be guaranteed even after a module update
- Effects during dynamic partial reconfiguration have to be considered to prevent an operational interruption of the system
- Radiation induced errors (SEU “bit-flip”) have to be considered
- **Isolate reconfigurable modules from host system !**

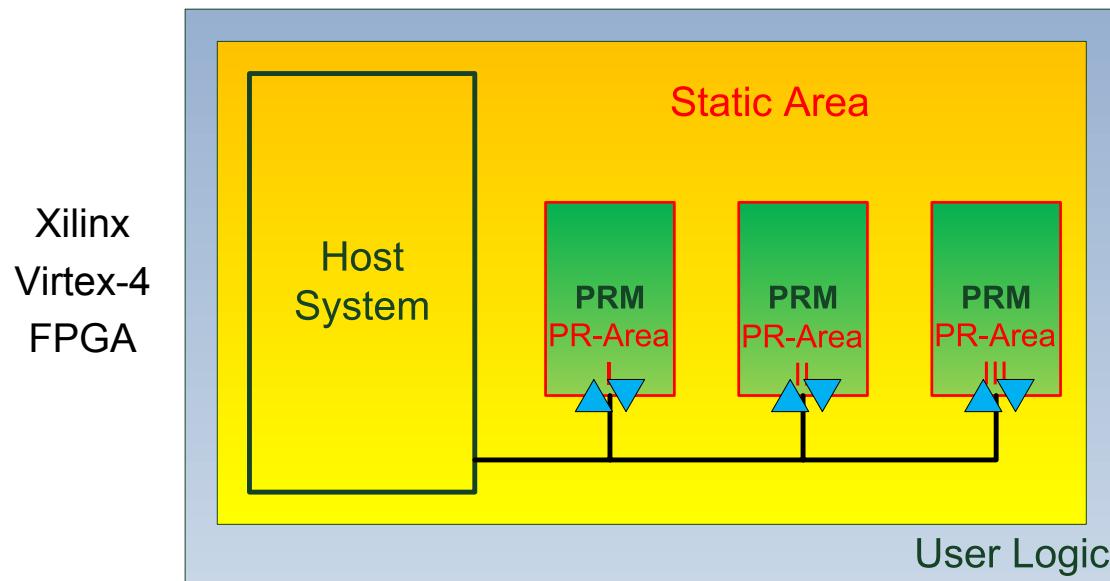


Bus structure based dynamic reconf. SoC



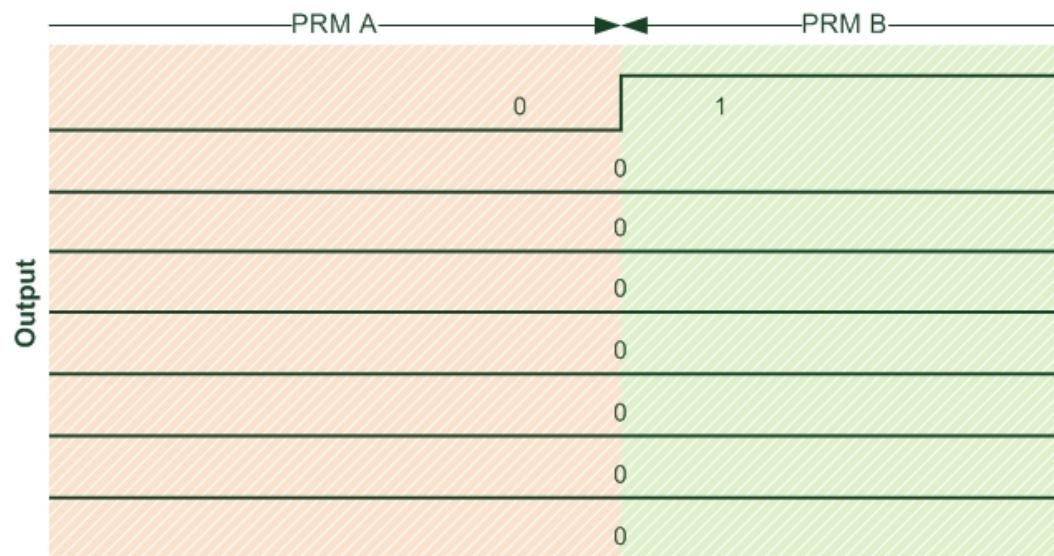
- Static Area → remains unchanged and stores all critical interfaces (CPU, to S/C)
- Partial Reconfigurable Areas (PR-Area) → can be updated during flight
- + Only the updated Module has to be qualified in a delta-qualification step

Bus structure based dynamic reconf. SoC



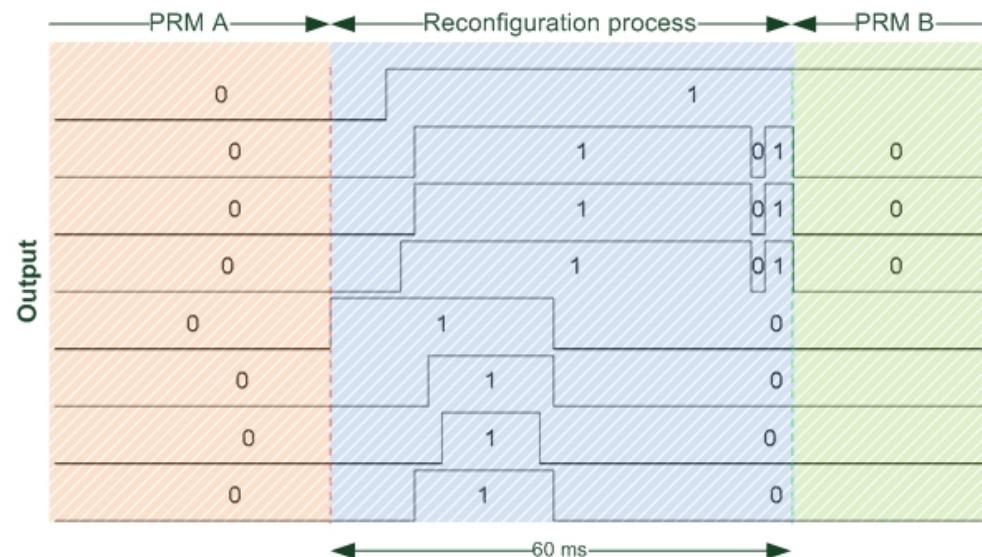
- Communication between static and PR-areas requires fixed routing
- Xilinx Bus-Macros suitable for bus standards (AMBA, Wishbone)

“Glitchless” dynamic reconfiguration



- Both modules in reset state:
 - PRM A output “0”
 - PRM B output “1”

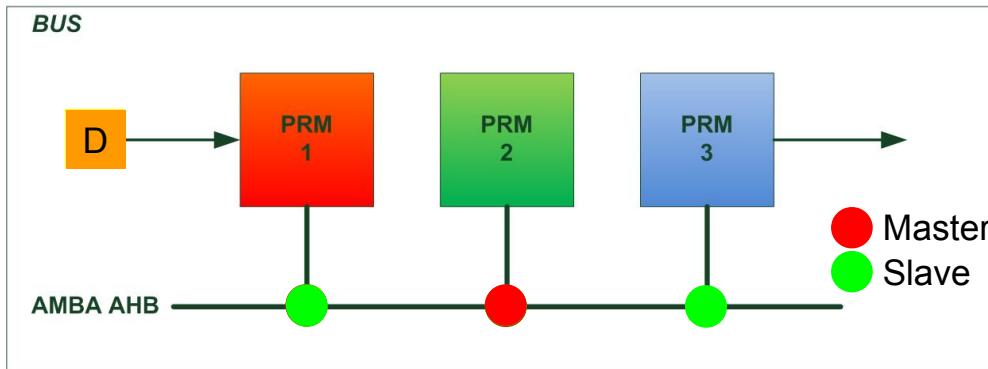
Xilinx dynamic reconfiguration behaviour



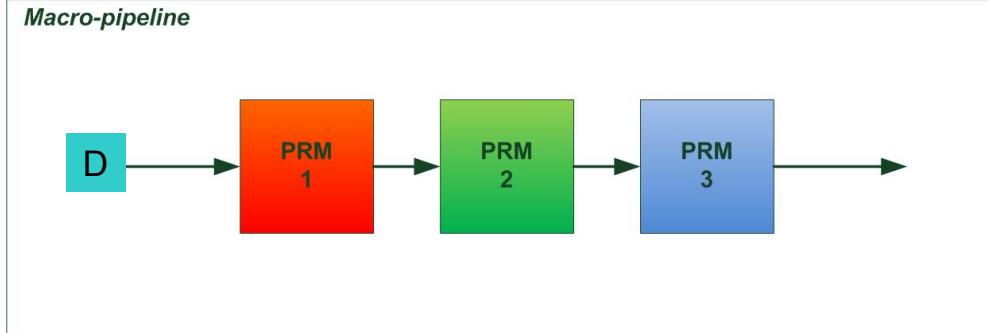
- Dynamic partial reconfiguration process does not have an explicit activation !
 - New “bits” become active as they are written
- **Unpredictable behaviour on the bus !**

Macro-pipeline

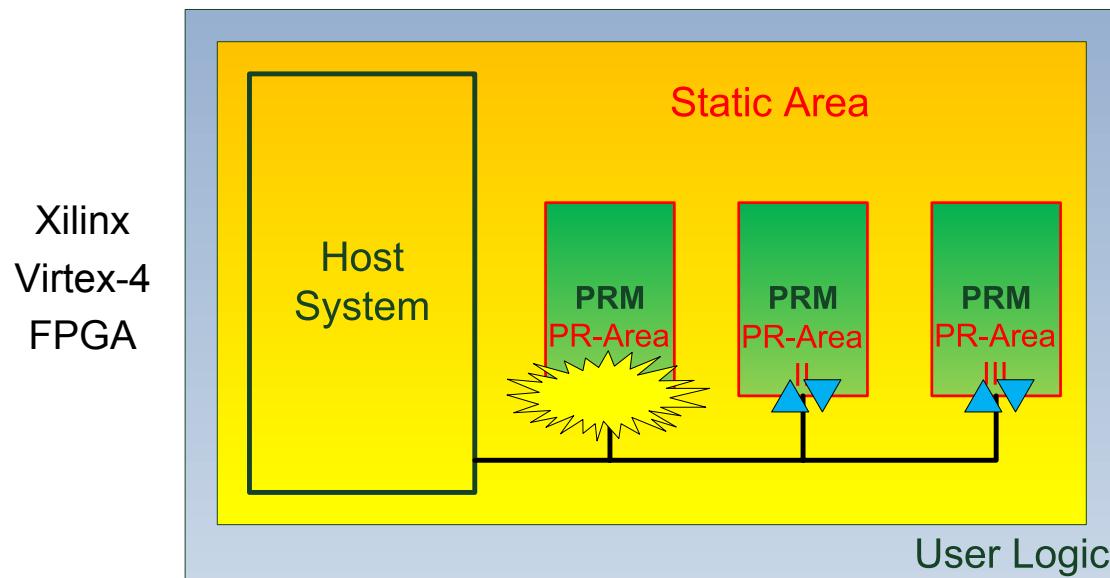
- **AMBA AHB**
 - Only one master has access to the bus at one time
 - Efficient bus arbitration necessary
 - Bus is bottleneck



- **Macro-pipeline**
 - Continuous data stream
 - High speed point-to-point connection
 - No schedule mechanism required



Bus structure based dynamic reconf. SoC



Limitation of bus structure:

- DPU macro-pipeline → Multi master, requires efficient bus arbitration
- SEU in PRM or dynamic reconfiguration process could block bus → stop system
- Failure tolerant bus structure (high efforts) necessary to guarantee data integrity
- No hot-plug ability



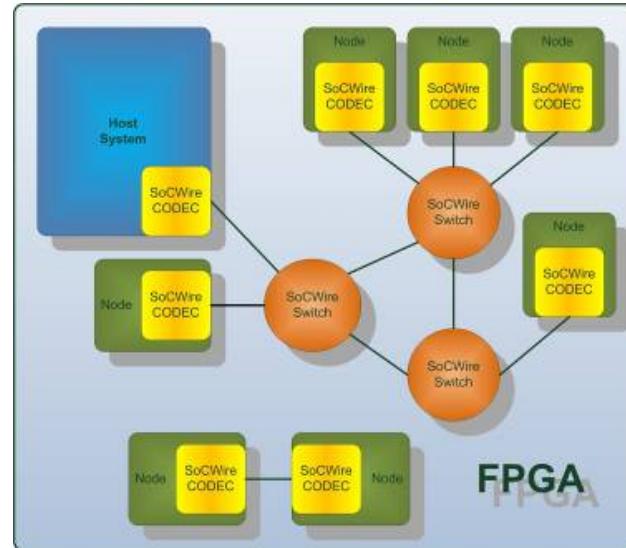
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Communication Architecture:

- Reconfigurable point-to-point communication
- Support of adaptive macro-pipeline
- High-speed data rate
- Hot-Plug ability to support dynamic reconfigurable modules
- Easy implementation with standard Xilinx Bus-Macros
- Logically separate PRMs from Host System



Network-on-Chip approach System-on-Chip Wire (SoCWire)



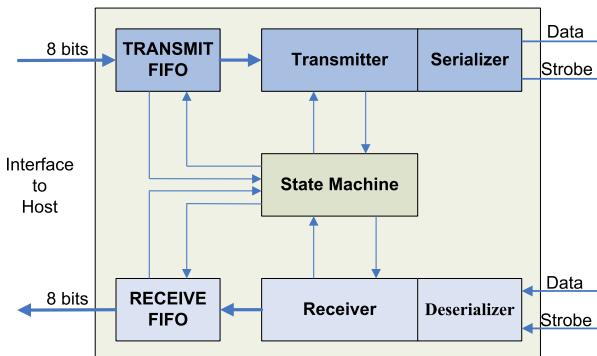
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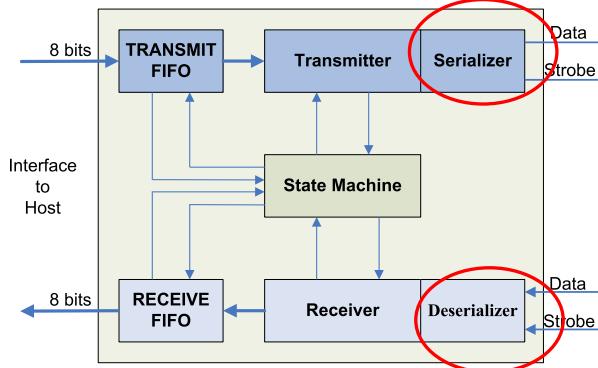
Spacewire

- Established standard in Space applications
- Serial interface
- Layered protocol
- Point-to-point links
- Bi-directional (full-duplex)
- Exchange Level Protocol
- + Link initialization
- + Flow control
- + Detection of link errors
- + Link error recovery
- + Low resource utilization
- Meets all requirements for a fault tolerant NoC



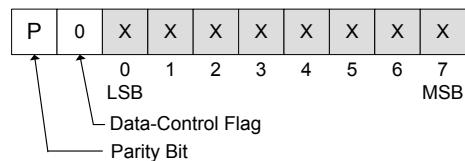
Spacewire

- SpaceWire uses Data-Strobe (DS) encoding
- Performance depends on:
 - Skew-Jitter
 - Technology



SpaceWire character level protocol

Data Characters



Control Characters

Control Characters

P | 1 | 0 | 0 *FCT Flow Control Token*

P | 1 | 0 | 1 *EOP Normal End of Packet*

P | 1 | 1 | 0 *EEP Exceptional End of Packet*

P | 1 | 1 | 1 *ESP Escape*

Control Code

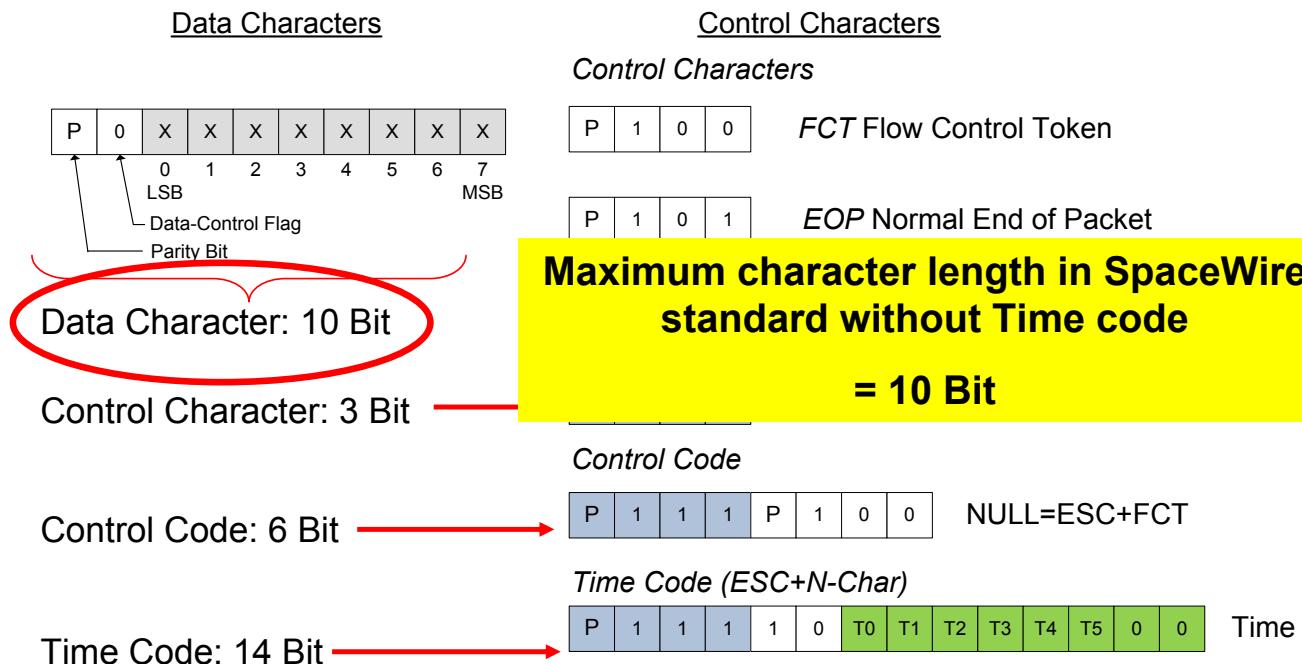
P | 1 | 1 | 1 | P | 1 | 0 | 0 *NULL=ESC+FCT*

Time Code (ESC+N-Char)

P | 1 | 1 | 1 | 1 | 0 | T0 | T1 | T2 | T3 | T4 | T5 | 0 | 0 Time

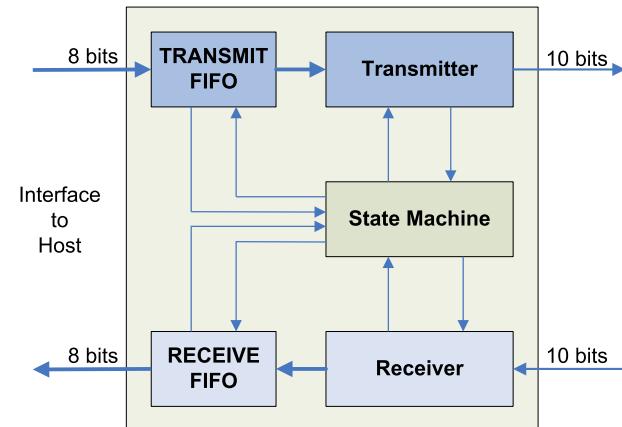


SpaceWire character level protocol

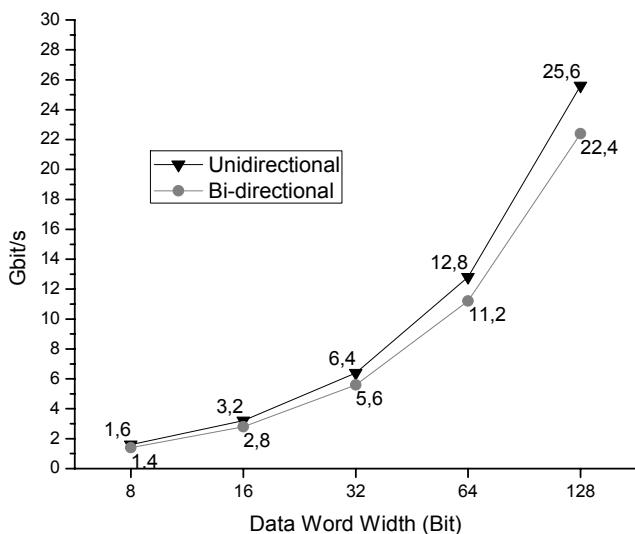


SoCWire CODEC

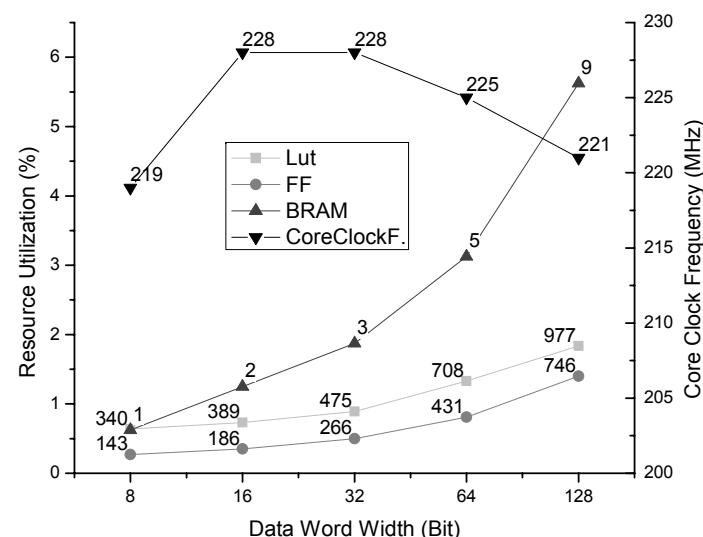
- SpaceWire uses Data-Strobe (DS) encoding
- Performance depends on:
 - Skew-Jitter
 - Technology
- On-chip environment → modified interface
 - 10 Bit parallel data transfer
 - Spacewire standard still supported without time code
 - Scalable data word width (8-128bit)



SoCWire CODEC Performance



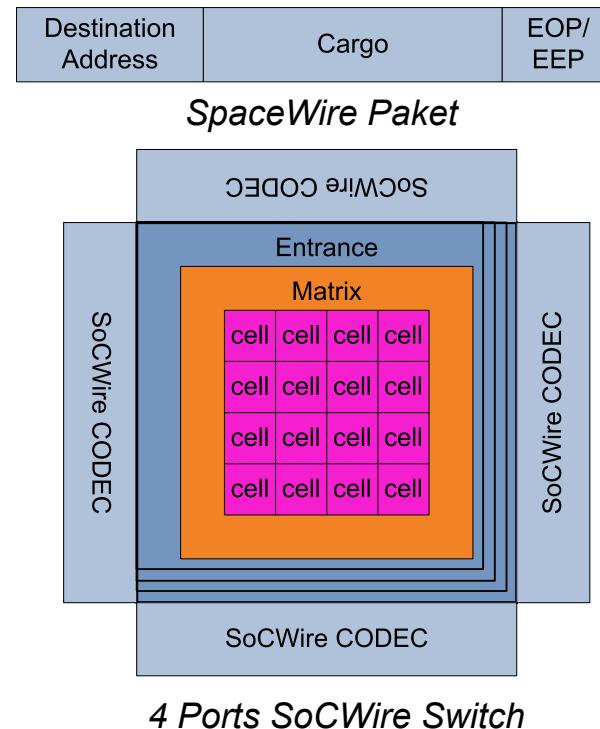
Data rates @ 200 MHz



Synthesis report
(Xilinx Virtex-4 LX60 -10)

SoCWire Switch

- SoCWire Switch
- Scalable data word width (8-128bit)
- 2 to 32 ports
- Direct port addressing and header deletion
- Wormhole Routing
- Entrance
- Entrance Modules = number of Ports
- Determines destination port
- Header deletion
- Matrix
- Cargo management
- connects the individual ports
- Cell
- Each cell represents connection between 2 ports
- 4 Ports = 16 Cells



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- SoCWire Switch 4 Ports synthesis report (Xilinx Virtex-4 LX60-10)

DWord Width	Max. f_{Core} (MHz)	Area	
		LUT	FlipFlops
8	190	1736	668
32	180	2540	1169

- Maximum data rate equivalent to SoCWire CODEC



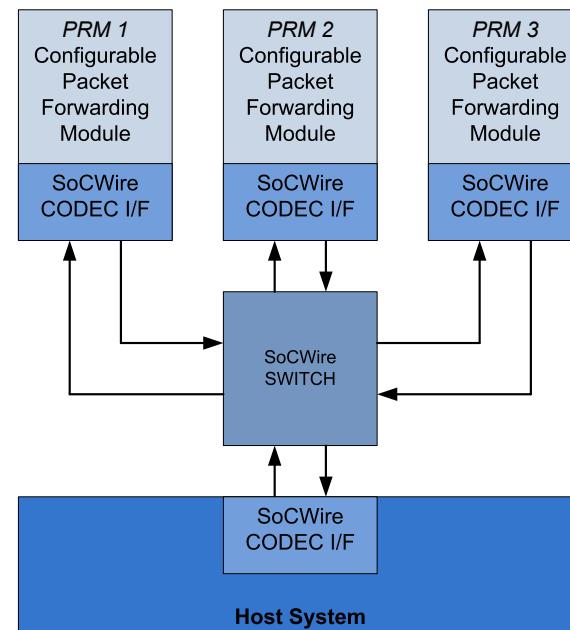
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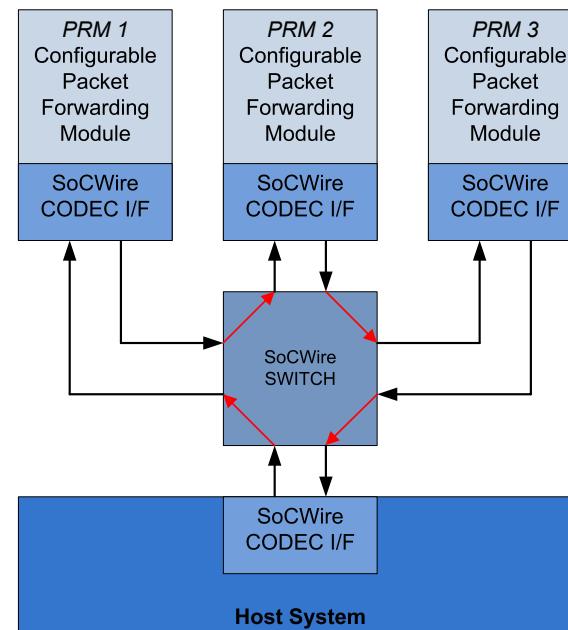
Test and Results

- Dynamic reconfigurable macro-pipeline system
- Static area:
 - SoCWire Switch
 - Host System
- Partial reconfigurable Areas
 - PRM 1,2,3
- PRMs: configurable packet forwarding Module
- 8 Bit data word width



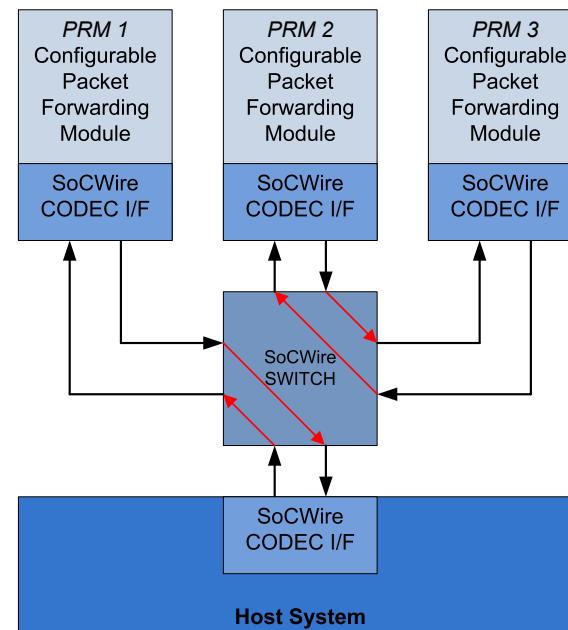
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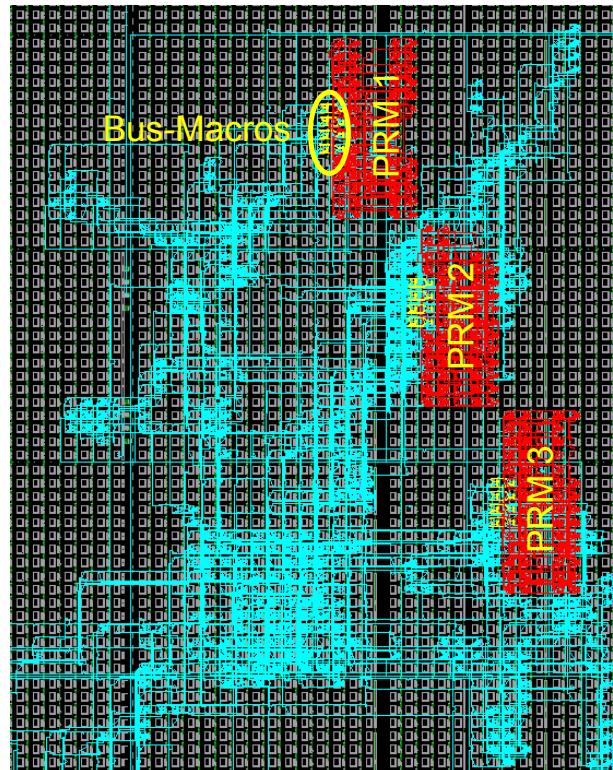
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Test and Results

- Xilinx Virtex-4 LX60-10
- Easy implementation with Xilinx standard Bus-Macros
- Static area distributed over FPGA
- Clock 100 MHz:
- Data rate macro-pipeline
- Unidirectional 800 Mbps
- Bi-directional 700 Mbps
- Dynamically reconfigured one PRM
- During reconfiguration process communication interrupted
- After reconfiguration, communication build up automatically without any external action

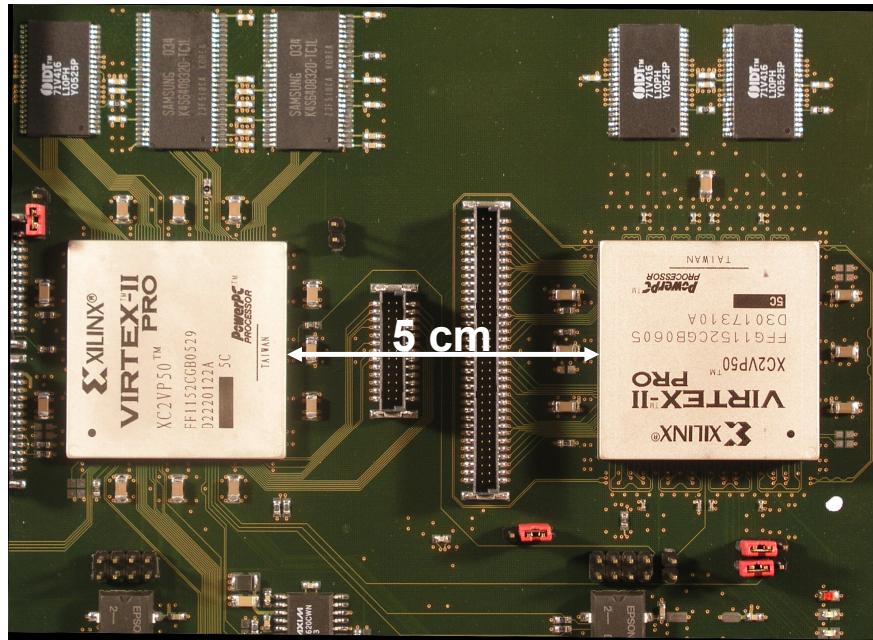


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Test and Results



- FPGA to FPGA communication
- SoCWire CODEC: 8 bit data word width @ 100 MHz → 700 Mbps (Bi-direct.)

SoCWire

- SoCWire
- Flexible high-speed communication architecture
- Support of adaptive macro-pipeline
- Hot-plug ability
- Suitable for dynamic reconfigurable systems
- SoCWire is distributed under an Open Source license

www.socwire.org

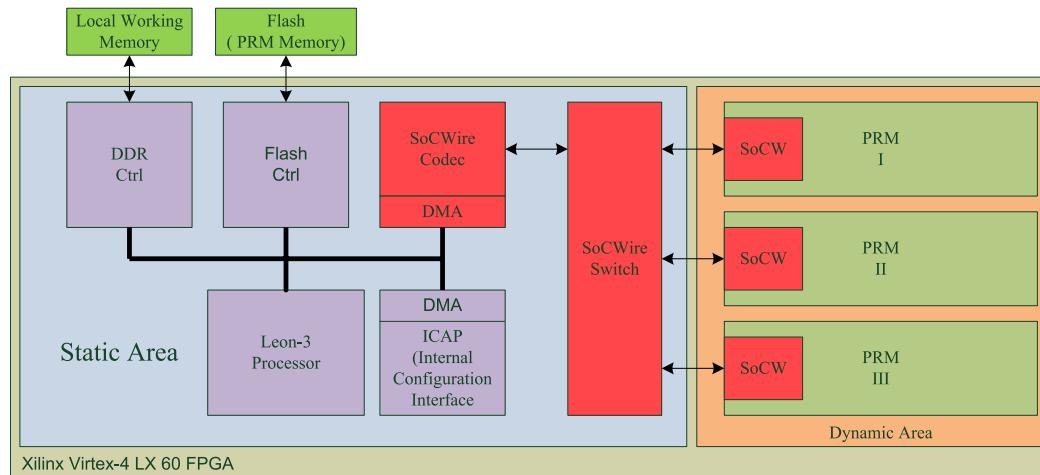


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Dynamic Reconfigurable Architecture



- GRLIB LEON-3 from Aeroflex Gaisler
- SoCWire with DMA capability on AMBA
- ICAP Module to access configuration memory
- 3 individual PRMs

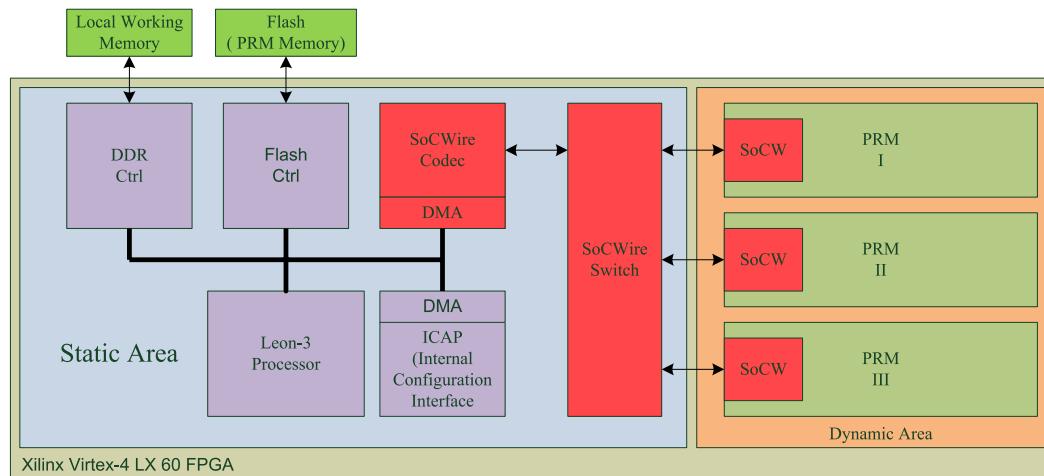


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Dynamic Reconfigurable Architecture



- SoCWire will be distributed within the GRLIB library of Aeroflex Gaisler

Conclusion

- Configurable SoC proven solution for scientific space applications
- Future space mission demand high performance on-board processing
- In-flight and Dynamic reconfigurability is a further enhancement to support update of hardware functions on-board
- Limitation of bus structure based dynamic reconfigurable SoC
- SoCWire
 - Flexible high-speed communication architecture
 - Support of adaptive macro-pipeline
 - Hot-plug ability
 - Suitable for dynamic reconfigurable systems
- Qualification of the system can be guaranteed
- + Isolates reconfigurable modules from the host system



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